

SUPPLEMENTAL TYPE CERTIFICATE

10056498 REV. 2

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

ATELIER CHABORD

**125 ROUTE DE BELLEGARDE
74330 EPAGNY
FRANCE**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.A.367

Type Certificate Holder: CEAPR

Type: DR-400

Model: DR400-180/ DR400-180 R
DR400-180S

Original STC Number: STC DGAC 45-SF-0001

Description of Design Change:

Add the new Duc propeller certified on DR 400/ 180 through STC 10059338 as NAC concerning the noise.

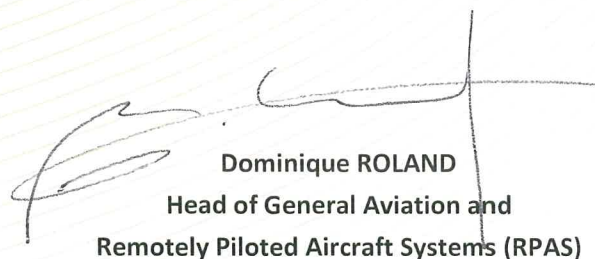
EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: 23.1529 and 23.1587 at CS23 Amdt2.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 05 December 2016



Dominique ROLAND
Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)

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The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

- Flight Manual Supplement SUPMV-DR480-16-01B dated 15/11/2016
or later revisions of the above listed documents approved by EASA.

- Technical dossier STC-DR480-16-01 Revision C dated 23/11/2016
- Noticemontage-dr480-16-01 Revision B dated 23/11/201

Limitations/Conditions:

The approval holder shall fulfill the obligations of Part 21, Paragraph 21A.109 or 21.A.451(b) (as applicable).

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -