

## SUPPLEMENTAL TYPE CERTIFICATE

**10048798 REV. 2**

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

### **ATELIER CHABORD**

**125 ROUTE DE BELLEGARDE  
74330 EPAGNY  
FRANCE**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Type Certificate Number:** EASA.A.367

**Type Certificate Holder:** C.E.A.P.R.

**Type:** DR 300/ DR 400 series

**Model:** DR 300/140, DR 340, DR 360

DR 400/140, DR 400/140 B

DR 400/160, DR 400/160 D

**Description of Design Change:**

STCDR460-2012-01 Rev. K dated 28 December 2016 - DR300/DR400 series Exhaust system and silencer.  
Reason for revision is to correct a typo mistake in Propeller P/N.

**EASA Certification Basis:**

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

See Continuation Sheet(s)

**For the European Aviation Safety Agency**

**Date of Issue:** 06 March 2017



**Dominique ROLAND**  
Head of General Aviation and  
Remotely Piloted Aircraft Systems (RPAS)

10017102  
SUPPLEMENTAL TYPE CERTIFICATE - 10048798 - REV. 2 - ATELIER CHABORD - 300699

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

**Associated Technical Documentation:**

AFM Supplement Ref. SUPMV-DR460-140-01C dated 28/02/2017

or later revisions of the above listed documents approved by EASA.

Document noticemontage-Dr460-14-01 Rev C dated 28/02/2016.

**Limitations/Conditions:**

The approval holder shall fulfill the obligations of Part 21, Paragraph 21A.109 or 21.A.451(b) (as applicable).

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -